# Integrated Control of Individual Scalars to Regulate Profiles and Improve MHD Stability in Tokamaks

A. Pajares<sup>1</sup>, E. Schuster<sup>1</sup>, K. Thome<sup>2</sup>, A. Welander<sup>2</sup>, J. Barr<sup>2</sup>, N. Eidietis<sup>2</sup>, D. Humphreys<sup>2</sup>



<sup>1</sup>Lehigh University, Bethlehem, Pennsylvania 18015, USA, <sup>2</sup>General Atomics, San Diego, California, USA E-mail: andres.pajares@lehigh.edu





### What do we mean by: "Integrated Control of ..."?

• We mean that the components of a control architecture work in an **interconnected fashion**, rather than working as isolated elements



Non-integrated architecture



Integrated architecture

In addition, an **integrated architecture** includes **supervisory and exception** handling (S&EH) algorithms and actuator manager(s) (more on this later)



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DIII-D

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What do we mean by: "Integrated Control of Individual What do we mean by: "... Improve MHD Stability ..."? Scalars to Regulate Profiles ..."?

 Controllability of a profile is sometimes limited. Instead, controlling associated scalars (e.g. volume-average) can be more attainable



Current profile:	<b>Rotation profile:</b>
– Central safety factor, $q_0$	<ul> <li>Volume-average</li> </ul>
- Edge safety factor, $q_e$	rotation, $\Omega_{\phi}$

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otation profile:	Pressure profile:
<ul> <li>Volume-average</li> </ul>	<ul> <li>Thermal stored</li> </ul>
rotation, $\Omega_{\phi}$	energy, W

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• Mode suppression by localized ECCD, which is planned for ITER [1]



 Algorithms to track rational surface  $(R_{q_{in}})$  with ECCD  $(R_{ECCD})$  developed in DIII-D [2]

 May need additional NBI/EC heating to achieve "pre-NTM" values [3], which modify  $q_0$ ,  $\Omega_{\phi}$ , and  $W \implies \mathbf{NTM}$  control coupled with scalars control

ye et al., *Control of neoclassical tearing modes in DIII-D*, Physics of Plasmas 9, 2051 (2002) Gunter et al., Neoclassical tearing modes on ASDEX Upgrade: improved scaling laws, high confinement at high  $\beta_N$  and new stabilization

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Summary of this Work: Test Integrated Architecture in Architecture has been Tested in Nonlinear Simulations Architecture has been Tested in Nonlinear Simulations

## DIII-D with Controllers + Actuator Manager + ONFR



- Actuator manager: uses controller commands and control priorities to calculate optimal actuator commands within physical saturation limits
- S&EH: Off-Normal Fault Response (ONFR) [4]. Switches control **priorities** in real time (e.g. use gyrotrons for NTM control vs  $W_{th}$  control)

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• COTSIM is a simulation code developed by the Lehigh University Plasma-Control Group [5] specially suited for control testing and tuning



using COTSIM (Control-Oriented Transport Simulator)



Simulations Test Architecture in High- $q_{min}$  Scenario to Achieve High Performance, NTM-Free Operation

- Plasma 1: simulated with experimental inputs from shot 172538
  - Except for reduced ECH, which results in 2/1 NTM development at 2.7 s
- Plasma 2: simulated with experimental inputs from shot 172538

- Except for slightly lower  $I_p$  (-0.05 MA)

Under Integrated Feedback (Turned On at 0.9 s), the Target Scalar Evolutions are Achieved in Simulations

### Color code: Plasma 1 (no feedback), Plasma 2 (target), Plasma 3 (feedback: comes on at 0.9 s)



NTM Suppression and the Inputs of the Target Plasma Simulation are Achieved under Feedback in Simulation

#### **Color code:** Plasma 1 (no feedback), Plasma 2 (target), Plasma 3 (feedback: comes on at 0.9 s)



- Maximum ECH simulated, which results in no NTM development
- Plasma 3: simulated with inputs determined in feedback
  - The plasma starts from the conditions and inputs of the first simulation
  - The goal is to achieve the scalar evolutions of plasma 2 using feedback
  - The feedback scheme **does not know the required inputs for plasma 2**

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## Experiments in DIII-D Test Architecture in High- $q_{min}$ Scenario Following Similar Ideas as in Simulation

- Plasma 1: using experimental inputs similar to shot 172538
  - Except for reduced ECH  $\leq 1$  MW
- **Target:** evolutions far from Plasma 1, but attainable
  - Slightly lower  $I_p$  ( $\approx$  -0.05 MA)
  - Ideally, no NTM development
- Plasma 3: inputs determined in feedback
  - The plasma starts from the inputs of the Plasma 1
  - The goal is to achieve the targets using feedback
  - The feedback scheme does not know the required inputs for the target

Feedback Architecture Tested in DIII-D (Turned On at 2 s) with less Actuation Capability than in Simulation

#### Color code: Plasma 1 (no feedback), Target, Plasma 3 (feedback: comes on at 2 s) Same scenario as in **–** - Target simulations, however: --No FB control (180390) — FB control (180397) - No counter-*I<sub>n</sub>* NBIs 0.8 (M) 0.6 $\implies$ no $\Omega_{\phi}$ control **8** de Limited off-axis NBIs M (210s in 2 s pulse) $\implies$ limited $q_0$ control - - Target 0.2 ---No FB control (180390) - ECH $\leq 1.5 \text{ MW} \implies$ —FB control (180397) no NTM suppression $\bigvee$ W + $q_e$ : good -No FB control (180390) - - Target regulation in feedback Ð $\cdot$ No FB control (180390) **–**FB control (180399) 3.5 (turned on at t > 2 s) — FB control (180399) $q_0$ : good regulation only if 210s on and S 2.5 until NTM shows up No NTM suppression. 1.5 but a delay in its appearance is seen (due to W control) t (s) t (s)

### FB Inputs Achieved Scalar Control and Delay in NTM Development, but No Significant NTM Suppression

Color code: Plasma 1 (no feedback), Plasma 3 (feedback: comes on at 2 s)







(2

### Conclusion & Future Work

- An integrated-control architecture has been successfully developed and implemented in the DIII-D PCS
  - **Preliminary** architecture  $\implies$  work needed to define final architecture
- It shows good performance in simulations and DIII-D experiments
  - This provides **initial validation** and encourages **further experimental tests**
- Future work may include:
  - Addition of **new actuators and controllers** (e.g. magnetic coils + shape control, gas puffing + pellet injectors + density control)
  - Integration of the architecture with other elements of the DIII-D PCS (e.g. integration with Proximity Control)
  - Testing in ITER-like scenarios using COTSIM



Appendix A: Actuator Manager Solves a Nonlinear, **Constrained Optimization Problem in Real Time** 

The actuator commands **u** are calculated by solving the following problem,

$$\begin{array}{cccc}
& \underset{\mathbf{u}}{\min} & \underbrace{\mathbf{s}^{T} \mathbf{Q} \mathbf{s}}_{\text{Metric for controller-request status}} & + \underbrace{\mathbf{u}^{T} R \mathbf{u}}_{\text{Metric for actuator use}} \\
& \text{subject to constraints} \\
& (1) \text{ Controller requests:} & \underbrace{f(\mathbf{u})}_{\text{nonlinear function}} & + \underbrace{\mathbf{s}}_{\text{slack variables}} & = \underbrace{\mathbf{u}_{\text{requests}}}_{\text{controller requests}} \\
& (2) \text{ Actuation limits:} & \mathbf{u} \subset \underbrace{u_{\text{limits}}}_{\text{Subset of feasible u}} \\
\end{array}$$

- Both Q and *R* prioritize **controller requests** and actuators, respectively
- The utility function f(u) is defined by the controllers in use, characterizing the relative satisfaction of **competing priorities** as a numerical quantity
- Some linear examples of f(u):

Total power =  $\sum_{i} u_i$ , Torque =  $u_{co-I_p} - u_{counter-I_p}$ , Actuator failure =  $u_i$  (= 0) DIII-D A. Pajares (LU Plasma Control Group) / IAEA FEC / May 10 - 15, 2021

Appendix B: Nonlinear Control Design for Catch-and-Subdue Power using Lyapunov Techniques

Island-width control with catch-and-subdue: start with the Modified Rutherford equation



assume the power of the pre-emptive ECCD clusters is set to the maximum available. By setting

$$j_{c\&s}(\hat{\rho}_{c\&s}^{max}) = \frac{K_P \frac{w^2 \tau_R j_{\phi}}{rL_q} + j_{BS}}{K_{c\&s}},$$
(2)

where  $K_P > 0$  is a design parameter, (1) becomes

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$$\frac{\tau_R}{r}\frac{dw}{dt} = \Delta'_0 r - \frac{\tau_R}{r}a_2 K_p w - a_2 \frac{j_{BS}}{j_\phi}\frac{L_q}{w}\frac{w_{marg}^2}{3w^2} - \frac{a_2 L_q}{j_\phi}\sum_k \frac{K_{EC,k}}{w} j_{EC,k}(\hat{\rho}_k^{max}).$$
(3)

It is assumed that the pre-emptive clusters provide a stabilizing effect, so the last term in (3) is non-positive, and

$$\frac{dw}{dt} < -a_2 K_p w \implies w < w_0 e^{-t/\tau}, \tag{4}$$

where  $w_0$  is the initial island width, and  $\tau \triangleq 1/(a_2 K_p)$  is the characteristic suppression time.  $P_{c\&s}^{req}$  is computed from (2) as

$$P_{c\&ss}^{req} = \frac{1}{j_{c\&ss}^{dep}(\hat{\rho}_{c\&ss}^{max})} \left[ \frac{n_e(\hat{\rho}_{c\&ss}^{max})}{T_e(\hat{\rho}_{c\&ss}^{max})} \frac{\frac{\tau_R}{r} \frac{K_{Pj}\phi}{L_q} w^2 + j_{BS}}{w \frac{F_c\&ss}{\delta_{c\&ss}^W} + K_{c\&ss}} - \sum_k j_{EC,k}^{dep}(\hat{\rho}_k^{max}) P_{EC,k} \right].$$
(5)  
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