Mission and Configuration Studies for a U.S. Sustained High-Power Density Tokamak Facility\*

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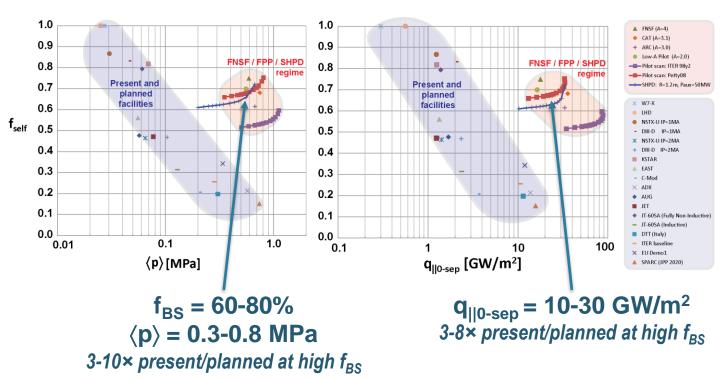
\*This work was supported by US DOE Contract No. DE-AC02-09CH11466 and the DOE Laboratory Directed Research and Development program.

# Overview

- U.S. fusion community supports Fusion Pilot Plant (FPP) mission:
  - Produce net electricity from fusion
  - Establish capability of high average power output
  - Safe production and handling of T, feasibility of a closed fuel cycle
- Reduced-cost tokamak FPP may be feasible through compactness
- Novel high-B, high-J magnets pursued by several groups
  - Standard aspect ratio (A~3), pulsed tokamak pursued by CFS
  - Aspect ratio A ≤ 2, non-inductive pursued by Tokamak Energy
- Compact + steady-state + high-power integration challenge
- Sustained high power density (SHPD) facility to address challenge

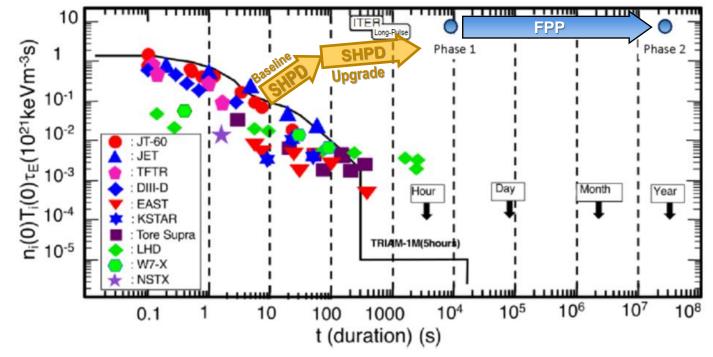
FPP parameters not yet accessible (simultaneously)

 Present and near-term planned facilities do not access the FPP regime of combined high self-driven current + high core plasma pressure + high divertor parallel heat flux



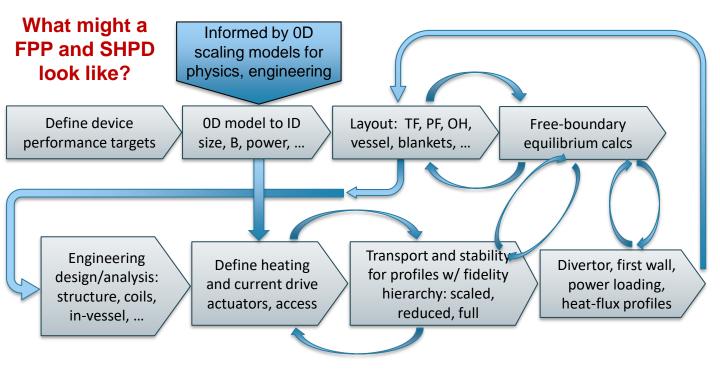
- Gap: 2-3 orders of magnitude in both pulse duration and  $nT\tau_{E}$
- Baseline SHPD device to narrow nT $\tau_{\rm E}$  gap, Upgrade to narrow  $\tau_{\rm pulse}$

Adapted from Fig. 4.2 of NASEM report "Bringing Fusion to the U.S. Grid" (2021)





# Systems, Scenario, and Device Configuration Modelling



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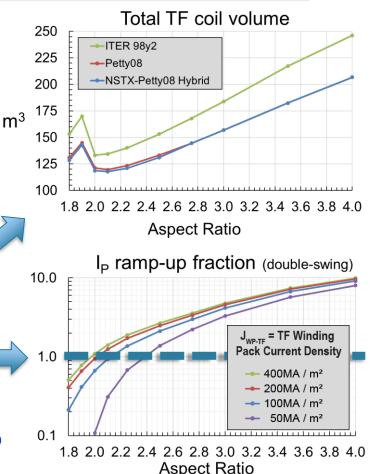
### Low-A reduces FPP magnet volume

#### **Pilot Plant example:**

- Net electric power = 100MWe
- 100% non-inductive current drive
- Tritium breeding ratio  $\approx 1$
- ReBCO TF lifetime = 10 FP-years

• B<sub>TF-max</sub> = 18T, J<sub>WP-TF</sub> = 50 MA/m<sup>2</sup>

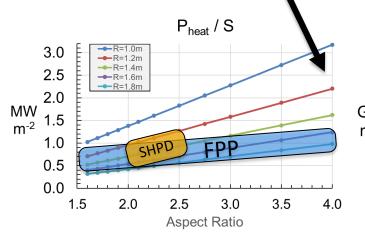
- A ≈ 2-2.4 reduces TF volume by factor of 1.5-2× vs. standard A
  May reduce TF cost (TBD)
- Low-A also reduces size, space available for central solenoid
  - Pulsed operation favors high A
  - A < 2  $\rightarrow$  partial non-inductive ramp

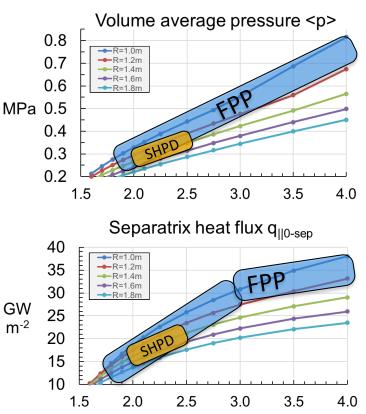


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- Average pressure
- Separatrix parallel heat flux
- Surface-average heat flux P<sub>heat</sub> / S
- High-A, small radius SHPD would have excessive P<sub>heat</sub> / S (3× FPP)





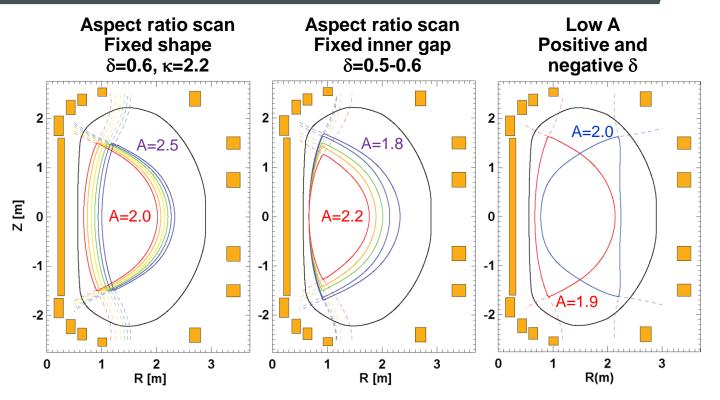
Aspect Ratio

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#### **Examples of shape flexibility with present SHPD coils**



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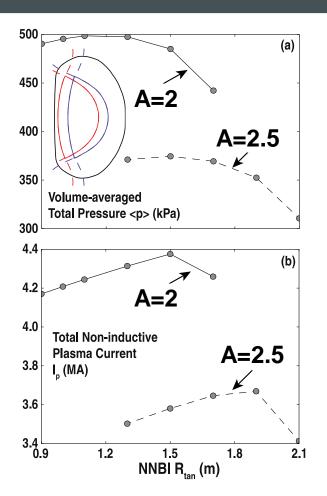


• Present PF set can access A=1.8-2.5 in single device

• Caution: Need to assess in-vessel component and exhaust compatibility

• Future: Explore long-leg / super-X / X-point-target at higher A=2.3-2.5

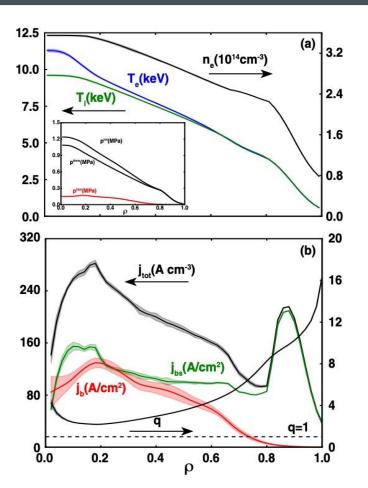
#### SHPD steady-state performance projections





- Model assumptions:
  - D-NNBI at 0.5 MeV, f<sub>G</sub> = 0.7-0.8
  - TRANSP, NUBEAM for H&CD
  - EPED1-NN for pedestal
  - GLF23 for core transport
- A=2.5 projections:
  - R<sub>0</sub> = 1.66m, B<sub>T</sub> = 4.4T, 20MW
  - = 0.37 MPa
    - Approx. ½ of FPP at A=2.5-3
- A=2.0 projections:
  - R<sub>0</sub> = 1.34m, B<sub>T</sub> = 5.4T, 20MW
  - = 0.5 MPa
    - Comparable to FPP at A=2.0

#### SHPD profile projections for A=2.0 scenario



- R<sub>0</sub>=1.34m, B<sub>T</sub>=5.4T
- P<sub>aux-NNBI</sub> = 50MW
- GLF23 + EPED1-NN + NSTXlike pedestal width scaling:
  - T(0) ≈ 10keV
  - I<sub>P</sub> = 4.5 MA (100% NI)
  - (p) = 0.54 MPa
  - $\beta_N = 3.7$ ,  $\beta_t = 4.8\%$ ,  $\beta_p = 1.9$
  - $f_G = 0.88$
  - q<sub>min</sub> = 2.2
  - f<sub>BS</sub> = 0.69
  - H98(y,2) = 1.65
  - Q<sub>DT</sub>(thermal) ~ 0.63

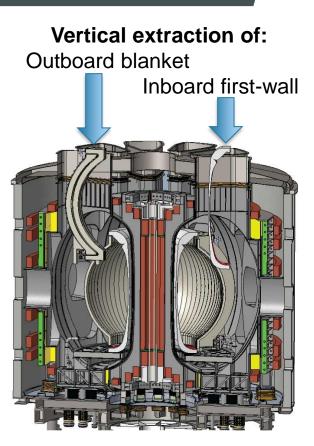


# **Example Device Configuration**

### Example engineering design concept

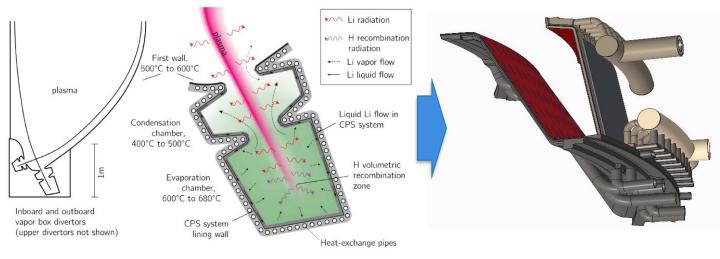
## **Example SHPD features**

- 10 HTS superconducting TF
- $B_{T0} = 5.5$  to 6.0T at  $R_0 = 1.2m$ depending on  $J_{WP}$  of TF
- Superconducting solenoid for  $I_P$  ramp-up, flux depends on  $J_{WP-TF}$
- Double null
- Outboard test blankets possible
- Vertical maintenance
- High-Z solid and liquid-metal walls





#### Example Li vapor box divertor concept / idea



Idea pioneered by R. Goldston

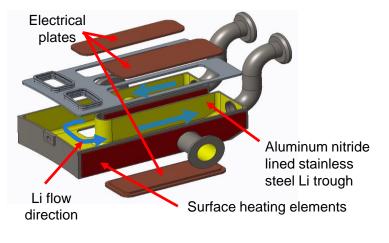
Jacob Schwartz – PhD Thesis September 2020 Princeton University

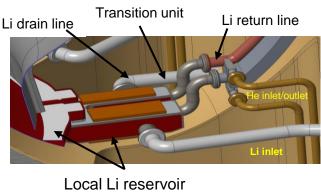
Continuing work by E. Emdee

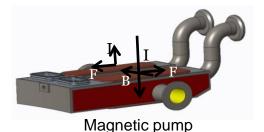
Vapor box module (w/o internal baffles) with lithium manifolds included Much more R&D to do!

#### Example Li magnetic pump system for fast flow







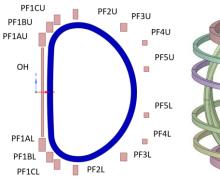


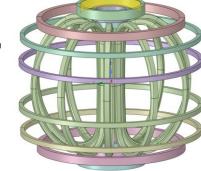
- 5-10 m/s Li flow velocity for 10 MW/m<sup>2</sup> convective heat removal
- Combinations of vapor box and fast-flow also being explored



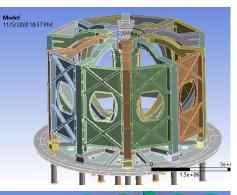
# **Initial Engineering Analysis**

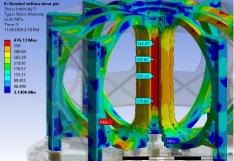


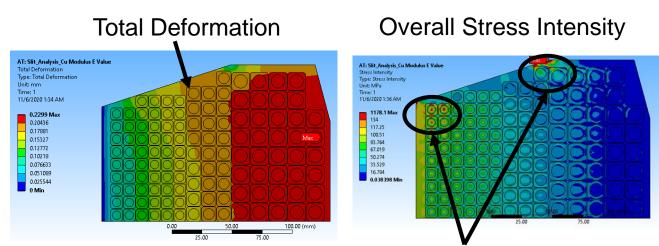




Electromagnetic structural load TF coils + casings + PF coil load Peak Stress= 380MPa is within allowables





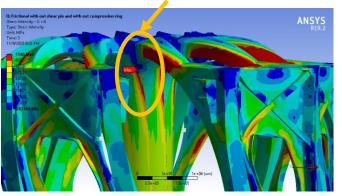


Max stress concentration areas in bundle corner regions

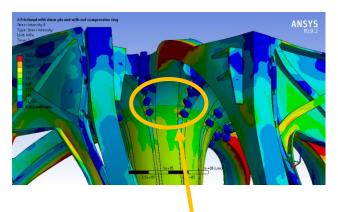
Deformation and stress through bulk of coil are within allowables, but some regions near casing corners need further optimization



#### **No shear pins:** Large sliding and gap opening TF-to-TF



With shear pins: Small sliding and gap opening TF-to-TF



1<sup>st</sup> shear pin region take max load

### Compression ring + shear-pins promising combination to reduce sliding and gaps to acceptable levels

- Present/near-term planned facilities will not access FPP regime
- Need dedicated facility (or FPP itself) to simultaneously explore:
  - High fraction of self-driven current
  - High core plasma pressure
  - High surface-average and divertor parallel heat flux
- R=1.4  $\pm$  0.2m, B = 4-6T, A = 2-2.5, P<sub>H&CD</sub>=50MW attractive for SHPD
- Systems studies and initial integrated predictive modelling indicate FPP regime should be accessible with the above SHPD parameters
- Initial device configuration and physics design integrates:
  - High current density and high B<sub>T</sub> toroidal field magnets
  - Lower aspect ratio / strong shaping to maximize f<sub>BS</sub> and pressure
  - Liquid metal systems (divertor, first wall, blankets) to prototype FPP
- Engineering calculations show pre-conceptual design is feasible